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MECHANICAL ENGINEERING DEPARTMENT

SCHOOL OF ENGINEERING

THE UNIVERSITY OF CONNECTICUT

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INTERIM PROGRESS REPORT for the Period November 1, 1963 through April 30, 1964

on

HEAT TRANSFER STUDIES OF VAPOR CONDENSING
AT HIGH VELOCITIES IN STRAIGHT TUBES

Reported by

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Mechanical Engineering Department University of Connecticut Storrs, Connecticut

May 1, 1964

Dr. T. L. K. Smull, Director Grants and Research Contracts NASA Washington 25, D. C.

Dear Sir:

The following interim report covers the research done under Contract No. NASA Code SC-NsG-204-62 for the period November 1, 1963 through April 30, 1964.

This work is a continuation of that performed under the same contract number for the period November 1, 1961 through October 31, 1963. In that contract period a series of non-linear, first order differential equations was derived to attempt to mathematically describe the high speed flow of a condensing vapor. Reference 1 of this report describes in detail these equations as well as their solution on a digital computer. The flow conditions used in the solution of the equations were based upon data for steam condensing in long, small diameter tubes, obtained from a series of tests (Refs. 2 and 3 of this report) performed at the University of Connecticut in the early 1950's. For several different entrance conditions and heat transfer rates it was possible to obtain a satisfactory correlation of the condensing heat transfer coefficients with the flow parameters. It was not possible, however, to obtain a suitable correlation of the two-phase friction factor.

It was decided that the difficulty lay in the interpretation of some of the test data, particularly the total pressure (vapor velocity) measurements. The first objective, therefore, of the present research is to obtain a more accurate set of data; secontly, to further develope the analytical treatment of the annular flow model.

This report covers the initial phase of this experimental work and describes the new test apparatus and changes in the analytical procedures.

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Charles H. Coogna, Jr.

Professor and Head of

Rechanical Engineering Department

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Description of Test Apparatus

The test apparatus has been designed with a different objective than the earlier equipment which was intended primarily to measure heat transfer rates in condensing flow. It is arranged primarily to determine pressure drop as a function of the heat transfer rate.

The test rig consists of a straight, concentric tube heat exchanger; the inner tube, twenty feet long with a 1.055 inch inside diameter and 0.035 inch wall, carrying the steam and the outer tube, nineteen feet long with a 2.009 inch inside diameter and a 0.058 inch wall, forming the cooling water annulus. Both tubes are commercial hard drawn copper. Figs. 1 and 2 show the overall layout and a typical cross section of the test apparatus. By reversing the direction of the cooling water flow, the test section can be operated in either a parallel or counter flow manner.

To obtain the desired measurements, the test section is instrumented in the following manner:

- i) Wall static pressure taps on the inner steam pipe formed by No. 60 drill holes connected to the outside manometers through hypodermic needle tubing.
- ii) Copper-constantan thermocouples to determine the temperature on the outer wall of the inner tube.
- iii) Copperaconstantan thermocouples in the cooling water annulus.
- iv) An exially traversing total pressure probe of the Dussourd type.

 (Ref. 4.)

Table I shows the axial location of these measurement points.

At the tube entrance are located a static pressure tap, and radially traversing total pressure and total temperature probes. From these

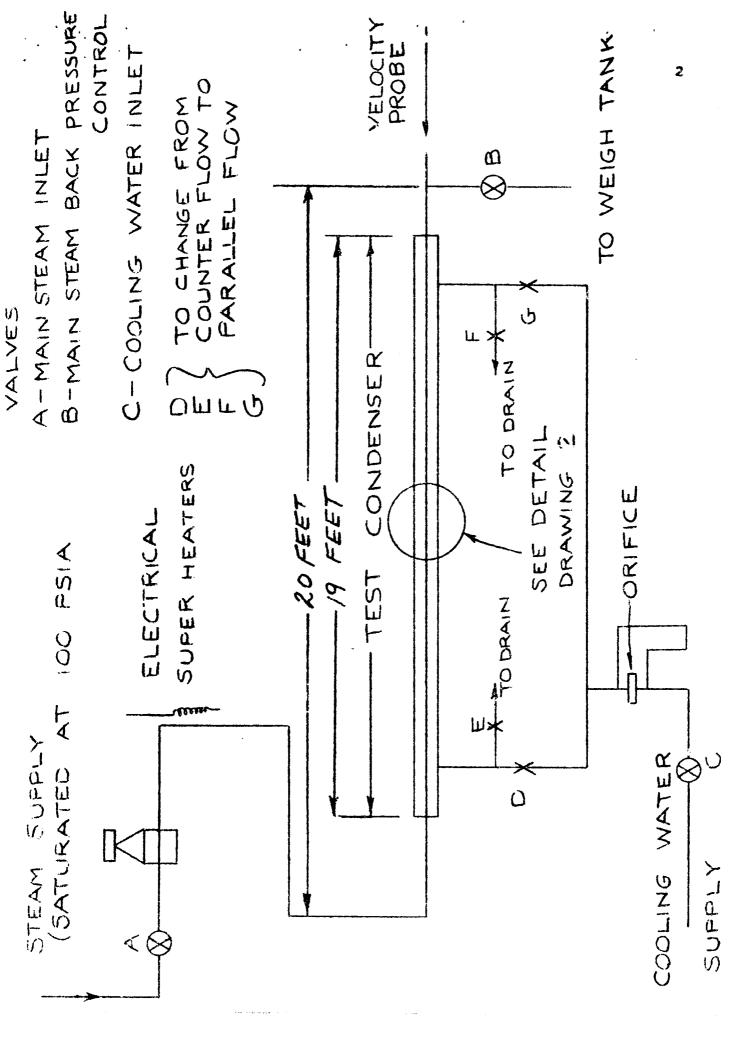
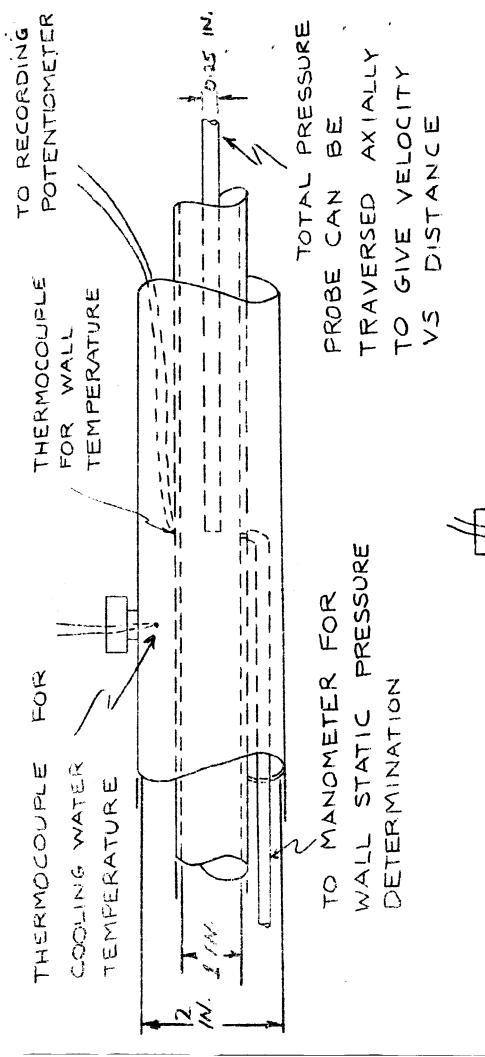


FIGURE 4



DETAIL DRAWING H

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Table I

Measurement Points

F - Indicates provision to make static pressure measurement

TC - Indicates provision to make temperature measurement

Position Yumber	Axial Fosition Ft. from Inlet	Wall of Inner Tube	Outer Wall of Inner Tube	Annulus Water Thermocouples
1		·		
1 2	1	P	TC	
3	1 74	P	TC	TC
Ĺ	2	P	TC	••
5	21	P	TC	TC
3 4 5 6 7 8 9	1½ 2 2½ 3 3½	P	TC	4 17
7	71	•	1	TC
ġ		P	TC	•
G	1.1	•	• •	TC
10	5	P	TC	10
11	1 61	,	1 19	TC
12	1 62		1	1 C
13	6.1		İ	TC
14	02	Р	TC	10
15	72	ŗ	10	mc.
16	4 4 5 5 5 6 6 7 7 2 8 8 8			TC
	8,		1	8 17)
17	82			TC
18	9 9 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	P	TC	
19	92		1	
20	10		l i	TC
21	103	_	i i	
22	11	P	TC	
23	12.3		1	TC
24	12		1	
25	125			TC.
2 6	13	P	TC	
27	136			T(
28	1	P	TC	
29	14.3			TC
30	15	P	TC	
	152	P	TC	3. C
32	16	P	TC	
33	163	P	TC	TC
31 32 33 34 35 36 37 38	177	•		- **
38	17/2	F	TC	TC
ã6	182	•	"	• •
37	18½	F	ge	TC
7 / 3 g	10	1		4 %
اب ال	1] [

At position number 1, provision is made to measure total and static pressure and to measure the total temperature of the flowing steam. Position I is located 3 inches from the later of the tube

measurements, the flow rate of the steam is computed and compared to that determined by weighing the condensate.

The total pressure probe of item iv is shown in Fig. 3. Briefly, it consists of a conventional total tube with the addition of static taps in the wall of the tube. These taps are located at distances from the tube mouth of 0.1, 0.2, 0.4, and 1.0 inches. When vapor with entrained liquid enters the mouth of the tube the vapor is quickly brought to rest, but the liquid, with its greater momentum, continues down the tube. A standard total pressure tube would read the combination of the liquid and vapor momentum change, giving an erroneous total pressure for the vapor alone. With the Dussourd-type probe, however, the static tap at 0.1 inch sees very nearly the true vapor stagnation pressure while the tap at 1.0 inch will, by manipulating the liquid interface in the total tube, see the combined pressure. The taps at 0.2 inches and 0.4 inches will give intermediate pressures. By plotting these pressures versus their axial location on the probe, the true vapor total pressure at the tube entrance can be obtained by extrapolation. This total pressure will be used to determine the steam velocity.

Test Facility Capabilities

d.) installed super heaters

As presently constructed, the test apparatus has the following maximum capabilities:

a 。)	cooling water flow rate	18,000 lbm/hr
ъ.)	regulated steam pressure	55 paia
c,)	steam flow rate	1500 lbm/hr

The experimental data can be obtained to within the following accuracies:

Steam flow rate	± 2%	Cooling water temperature	± 1%
Water flow rate	<u>+</u> 2%	Static pressure	± 23
Tube wall temperature	+ 4%	Dynamic pressure	+ 2%

25 k.w.

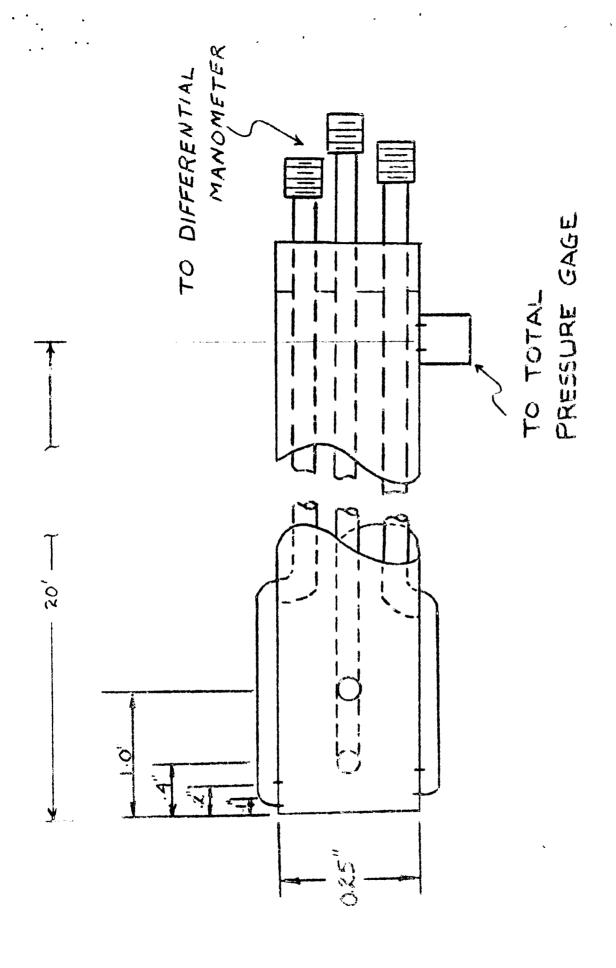


FIGURE DEPOSITION FRANCE

Procedure

Briefly, the test procedure is as follows: The manometers and lines filled with water and balanced; cooling water flow pattern determined and flow initiated; steam flow turned on and equilibrium flow rate established; velocity probe inserted to most forward position and then withdrawn, readings of total pressure made at prescribed axial positions; static pressure readings made with probe in and probe out.

The data obtained from the test apparatus is used as input for a data reduction program for the IBM 7040 computer. The information will be inserted in the form as recorded from the tests; i.e., pressures in psig and inches of mercury, temperatures in ^oF and flow rates in lbm/hr. This input will then be converted to consistent units and the data processed to produce as output the static and dynamic pressures in psia, tube wall and cooling water temperatures in ^oF, vapor and liquid velocities, the physical properties of saturated steam and saturated liquid such as enthalpy, density and entropy at each measuring point, and the condensing heat transfer coefficient in Stu/hr ft² ^oF. Tables II and III illustrate the input and output form.

Steam Table Computer Program

Appendix A of Ref. 1 describes a method of predicting steam properties from entrance conditions and local properties. However, this method coes not give the degree of accuracy now desired so a method making use of a computer sub-routine (Refs. 5 and 6) for the steam table properties was developed. This sub-routine determines enthalpy, entropy and specific volume as a function of pressure and temperature in the superheat region; to the saturated region, the vapor enthalpy, entropy and specific volume are found as functions of pressure; the saturated liquid enthalpy, entropy and specific volume are given

Input Data

Data Reduction Program

NASA Project

Nechanical Engineering Department

cions		Cooling Water Temp.	(Jo)
Cooling Water Conditions	Flow Rate (lbm/hr) Type Flow	Outside Wall Temp.	(^O F)
		Stag, Fressure	(in Hg) (psia)
Stagnation Chamber	Pressure (ps1g) Temparature (°F)	Static Fressure	(in Hg) (psia)
(y)		Distance.	(L

Output Data

Data Reduction Program

NASA Project

Mechanical Engineering Department

Cooling Water Conditions

Stagnation Chamber

Surface Coefficient (Btu/hr - ft² %) (π^2) (π^2) Area Area Flow Rate (15m/hr) Flow Rate (1bm/hr) Flow Rate (lbm/hr) Type Flow Inside Wall Temp. Velocity (tt/sec)Velocity (ft/80c) Saturated Liquid Properties Saturated Vapor Properties Cooling Water Properties (1bm/ft³) $(1bm/rt^3)$ Density Density Outside Wall Temp. Temp. Temp. (°F) (oF) Stat. Pres. Stat. Pres. (psia) (psia) Heat Rate (Btu/hr) Stag, Pres, Stag, Pres, Pressure (psia) Temperature (PF) (peia) (psia) Temp。 (P) Actual Dimensionless Actual Dimensionless Dimensionless Distance Distance Distance Actual (£ (rt) (£

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as functions of temperature and temperature and pressure are found as functions of each other on the saturation line.

A second sub-routine has been developed that calculates the properties of steam for a constant entropy process, using the steam table sub-routine. Valid results are obtained in the wet, saturated, and superheated regions. Given the properties at one point and the pressure at a second point, the routine will calculate the remaining properties at both points assuming an isentropic process.

Projected Work

During the next period, the test work previously described will be continued. The data processing and the computer program for the solution of the flow equations will be under continual revision. Using data from the existing test section, a method involving the use of successively smaller finite difference equations will be used to calculate a local wall friction factor.

This computation is based on the momentum equation and requires an accurate knowledge of the vapor velocity.

The problem of non-condensables in the steam is being investigated and some experimental apparatus obtained to gain quantitative information of air entrainment. This is a quite difficult problem and as of this date, no quantitative data has been obtained. Further work in this area is contemplated.

Additional Apparatus to be Constructed

Additional experimental apparatus will be constructed. It is expected that an annular heat exchanger of the same radial dimensions as the existing test section but approximately two feet long will be built. This section will be instrumented with electrical resistance probes to measure liquid

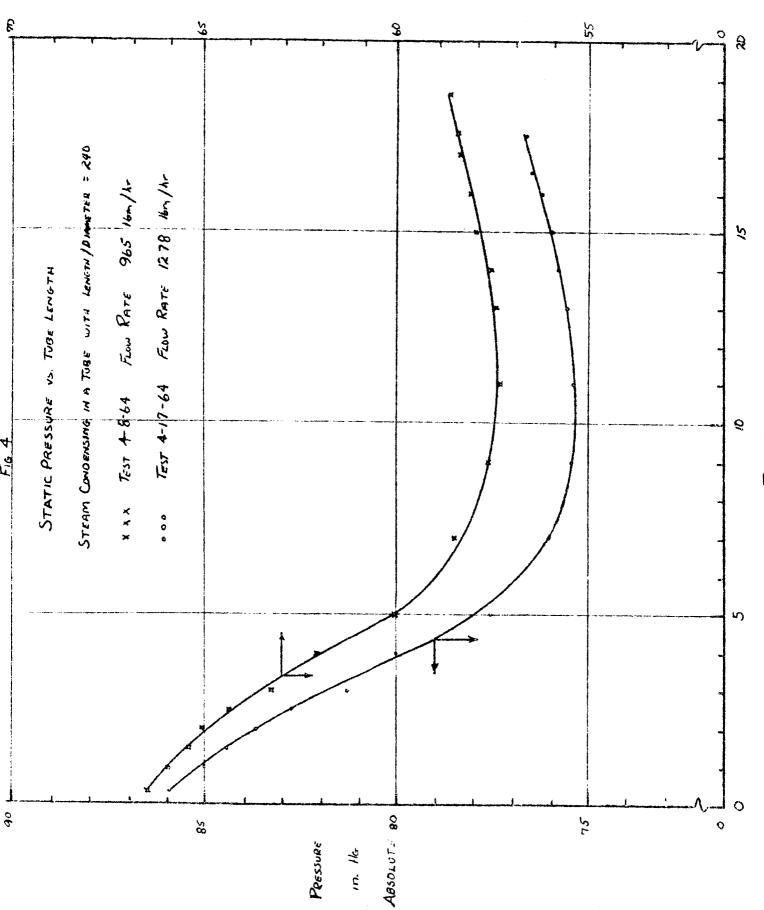
layer thickness, velocity probes to traverse the vapor flow in the radial direction and liquid layer velocity probes. The cooling water jacket will be closely monitored so that accurate heat transfer rates can be determined. The above data will be used in an attempt to develops a successful correlation of the local wall friction factor with the flow parameters and liquid layer thickness for condensing steam.

Another test section will be constructed utilizing glass in part of its length. Flow visualization studies will be made of the effect of probes inserted in the stream as well as studies of flow patterns under various pressure gradients.

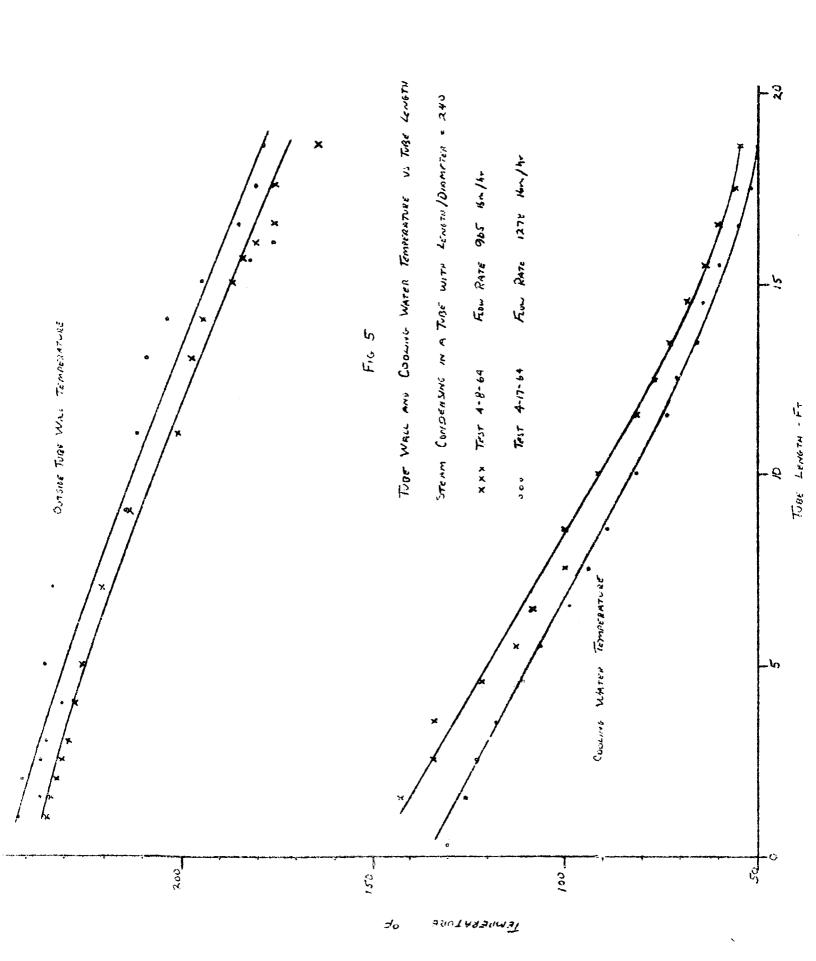
The results of the experimentation discussed above will be used to gain a broader insight of the mechanisms of condensing flow. With this additional knowledge, if need be, a better mathematical model can be developed and a more meaningful solution, based upon physical principles, made of the equations.

Results

The greater part of the period covered in this report was spent in the design, construction and de-bugging of the test apparatus. A complete set of test data has not yet been obtained although several runs utilizing the velocity pressure probe have been made. These runs were made, however, in the interest of developing the test procedures rather than gathering cata. Static pressure, tube wall and cooling water temperature data are available and Figs. 4 and 5 show two typical sets of these data.



TUBE LENGTH FT



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